Statistical Analysis of Atmospheric Flight Gust Loads Analysis Data

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Abstract

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Contents

Abstı	ract	1	
Nom	enclature	5	
1.	Introduction	7	
2.	Background	9	
3.	Characterization of Gust-Induced Loads	11	
4.	Determining the 0.997 Load Enclosure	13	
5.	The Bootstrap Procedure	15	
6.	Nonparametric Procedure	17	
7.	Conclusions	21	
Refe	References		

Figures

1,	Typical distribution of gust-induced load for a heavy lift launch vehicle	11
2.	Heavy lift launch vehicle gust-induced loads normal probability plot	12
3.	Gamma probability plot for the data shown in Fig. 2	12
4.	This plot demonstrates the normalizing transformation	17
5.	Nonparametric procedure for tolerance bounds	19
	Table	
1.	Coverage Probability of Bootstrapped 0.997/0.90 Gamma Tolerance Intervals	15

Nomenclature

\boldsymbol{C}	confidence level
E	load enclosure
\overline{F}_X or G_X	cumulative distribution function of X
F_{z}	cumulative distribution function of a standard normal random variable
f_X	probability density function of X
n	sample size
q_p^X	p-quantile of X
S	sample standard deviation
T_X	tolerance bound for X
$t_{n,\delta,p}$	p-quantile of non-central t distribution with n degrees of freedom and non-centrality
	parameter δ
\boldsymbol{X}	random variable
\bar{x}	sample mean
Z	standard normal random variable
z_p	standard normal p-quantile
р	$\mathbf{r} = \mathbf{r} \cdot \mathbf{r} \cdot \mathbf{r}$

1. Introduction

Gust loads analyses are performed to establish launch vehicle and space vehicle loads due to turbulence encountered during atmospheric flight. Historically, launch vehicle loads have been calculated by applying a synthetic gust profile whose amplitude, wavelength, and shape were selected to induce loads of a desired magnitude. In Ref. 3, a new Monte Carlo analysis approach that uses the turbulent component of measured wind profiles is presented. The approach requires statistical analysis of extensive loads analysis results. This paper presents the statistical analyses performed to establish a suitable distribution for gust loads, so that tolerance bounds on these loads could be calculated.

The objective of this work was to characterize the statistical distribution of the gust loads data in Ref. 3, and to determine the 99.7 percent load enclosure. The 99.7 percent load enclosure is the "3-sigma" load enclosure for data having a normal, or Gaussian, distribution; i.e., 99.7 percent of normal data lies within three standard deviations of the mean of the distribution. For data with statistical distributions other than normal, however, "3-sigma" is not synonymous with 99.7 percent coverage, and the 99.7 percent enclosure may be significantly different from the three-sigma enclosure. Because of the uncertainty in the parameters of the statistical distribution, the 99.7 percent enclosure is presented as a 90 percent upper confidence bound on the 99.7 percent enclosure. In statistical terminology, a 0.997/0.90 upper tolerance bound for the data is determined. Two tolerance bound procedures are described: one that assumes a gamma distribution for the gust loads and is most suitable for small sample sizes, and another that makes no distributional assumptions, but is useful only if the sample size is sufficiently large.

2. Background

The application and theory of statistical tolerance bounds, where the derivation of tolerance bounds for normal distributions, as well as for certain other distributions, is discussed. Tolerance bounds are derived for the generalized gamma distribution, a class of statistical distributions that includes the exponential, two-parameter gamma, and Weibull distributions. The procedure presented in Ref. 6, however, requires that some parameters of the distribution be known in order to calculate a tolerance bound. This procedure is not applicable to this work, since all parameters of the gust-induced loads distribution need to be estimated from the data.

3. Characterization of Gust-Induced Loads

Gusts, which for the purposes of this discussion are defined as the non-persistent, relatively short wavelength components of the winds⁷ that a launch vehicle will encounter during flight through the atmosphere, induce loads that have been approximated by a gamma distribution.^{8,9}

The distribution of a typical set of gust loads in Ref. 3, is shown in Fig. 1. The asymmetry of the data suggests that modeling the data with the normal distribution, and calculating the load enclosure using three-sigma limits could significantly underpredict the true 99.7 percent enclosure.

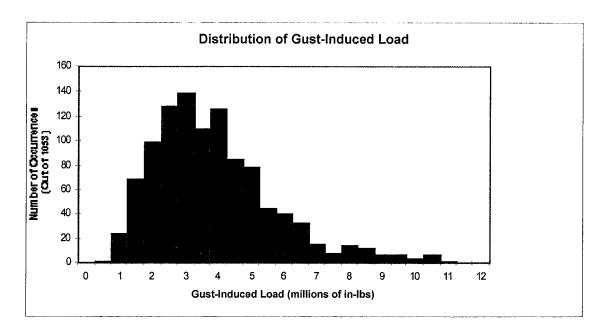


Figure 1. Typical distribution of gust-induced load for a heavy lift launch vehicle.³

This histogram has the long right tail typical of a gamma distribution.

Probability plots are a useful tool for assessing the distribution of data. A probability plot compares the observed data to what would be expected if the data had a particular distribution. If data have a normal distribution, the data will tend to lie on a straight line on a normal probability plot. Probability plots for other distributions can likewise be constructed, and data with a given statistical distribution will tend to lie on a straight line on the corresponding probability plot.

Normal and gamma probability plots for a typical set of gust load data in Ref. 3, are shown in Figs. 2 and 3. In order to create the gamma probability plot, an estimate of the shape parameter of the gamma distribution was necessary; the maximum likelihood estimate of the parameter was used in this plot. Superimposed on each plot is a least-squares, best-fit straight line. These plots indicate that modeling the gust-induced load with a normal distribution is inappropriate, and a gamma distribution models these loads fairly well.

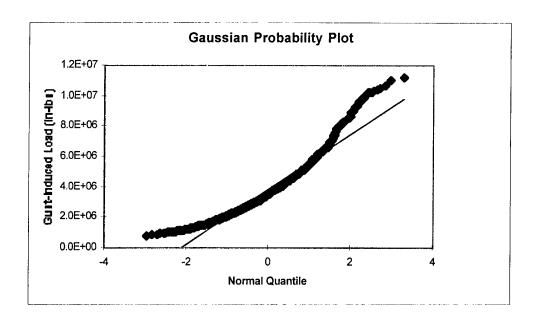


Figure 2. Heavy lift launch vehicle gust-induced loads³ normal probability plot.

Nonlinearity of data demonstrates the inappropriateness of modeling these gust-induced loads using a normal distribution.

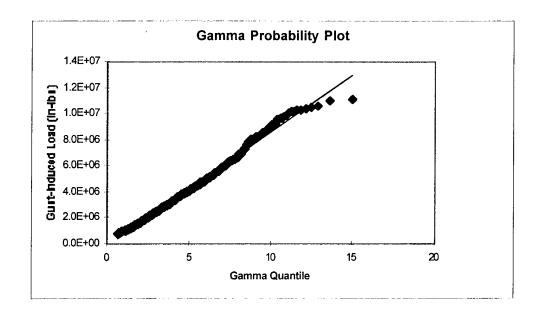


Figure 3. Gamma probability plot for the data shown in Fig. 2. A straight line fits the data well, suggesting that a gamma distribution is an appropriate model for these gust-induced loads.

4. Determining the 0.997 Load Enclosure

If the statistical distribution of the gust loads and the parameters of that distribution were known, it is straightforward to find a 0.997 load enclosure by integration. For normal data, a standard normal probability table can be used, with the result that 99.7 percent of the data lies within three standard deviations of the mean. For a general probability density, f_X , the 0.997 load enclosure is the value of E satisfying

$$\int_{-\infty}^{E} f_X(x) dx = 0.997.$$

If the true distribution is unknown (although the form of the distribution may be known, the parameters may not be), uncertainty in the load enclosure is introduced, and the 0.997 load enclosure can only be estimated. It is possible to place an upper confidence bound on this estimate, resulting in a one-sided tolerance bound on the load enclosure. In this work, the goal is to find a 0.997/0.90 tolerance bound; i.e., a 90-percent upper confidence bound on the 0.997 load enclosure.

For normal data, a p/C (e.g., 0.997/0.90) tolerance bound is of the form $\bar{x} + ks$, where \bar{x} is the sample mean, s is the sample standard deviation, and k is based on the non-central t distribution:

$$k=\frac{t_{n-1,\sqrt{n}z_p,1-C}}{\sqrt{n}},$$

where n is the sample size, $\sqrt{n}z_p$ is the noncentrality parameter of the distribution, and z_p is the standard normal p-quantile (e.g., $z_{0.975} = 1.96$). Tables for values of k are tabulated in Ref. 5, or can be readily computed.

Tolerance bounds for data other than normal data can be more difficult to determine. For gamma data, two procedures are described. The first method, utilizing a bootstrap statistical procedure, is useful for sample sizes under 1000 or so. The second procedure, useful for sample sizes greater than about 1000, is based on a normal transformation of the data.

5. The Bootstrap Procedure

The bootstrapping procedure is a resampling procedure that is useful for determining confidence bounds on estimates. On 10.11 Both nonparametric and parametric bootstrap procedures exist; a parametric procedure is used here to determine tolerance bounds for data having a two-parameter gamma distribution. Reference 10 employs the BC_a parametric bootstrap. The bootstrap, and the BC_a method, in particular, have nice asymptotic properties, but in order to demonstrate the effectiveness of this procedure for estimating percentiles of gamma data based on relatively small samples, Monte Carlo simulations were performed. For sample sizes of 10, 30, and 100, and gamma shape parameters of 1.5, 3, 5, and 10 (the results are independent of the scale parameter of the gamma distribution), the coverage probabilities of the BC_a tolerance bounds were estimated. Table 1 summarizes the results of the Monte Carlo simulation. The table entries are the proportion of 1000 bootstrap 0.997/0.90 tolerance intervals that covered the true 0.997 quantile of a gamma distribution with given shape parameter and given sample size. In each case, approximately 90 percent of the bootstrapped tolerance intervals include the true 0.997 quantile of the distribution, indicating that the BC_a tolerance bound results in accurate tolerance bounds.

Coverage Probability of Bootstrapped 0.997/0.90 Gamma
 Tolerance Intervals. Table Entries are the Proportion of 1000
 Bootstrap 0.997/0.90 Tolerance Intervals that Covered the True
 0.997 Quantile of a Gamma Distribution with Given Shape Parameter and Given Sample Size. All Entries are Near What is
 Expected: 0.90.

	Sample Size				
		10	30	100	
Gamma	1.5	0.87	0.90	0.92	
Shape	3.0	0.86	0.90	0.92	
Parameter	5.0	0.85	0.90	0.93	
	10.0	0.89	0.90	0.90	

Drawbacks to the parametric bootstrap procedure for calculating tolerance bounds include the following: 1) It is computer-intensive, taking a few minutes to perform for sample sizes on the order of 1000; and, 2) It is sensitive to the assumption of a gamma distribution. The second drawback cannot be avoided by using a nonparametric bootstrap; it is not possible to estimate the 0.997 enclosure of a small set of data without making assumptions about the distribution. The first drawback is less of an issue.

Applying this method to the data shown in Fig. 1, we obtain a 0.997/0.90 tolerance bound of 10,960,000 inch-pounds.

6. Nonparametric Procedure

A nonparametric method of calculating a tolerance bound can be used when the sample size is sufficiently large. This method relies on a standard result of probability theory that for a continuous probability distribution, a nondecreasing function exists that, when applied to the data, transforms the data to a normal distribution: If G is the cumulative distribution function of a continuous random variable X, and F_z is the standard normal cumulative distribution function, then f(X) has a standard normal distribution, where $f(\cdot) = F_z^{-1}[G(\cdot)]$. See Fig. 4 for a graphical explanation of this result, which follows from the *Probability Integral Transformation*. This result is useful for estimating tolerance bounds because if the transforming function, f, can be accurately estimated, tolerance bounds for non-normal data can be calculated using normal tolerance bounds.

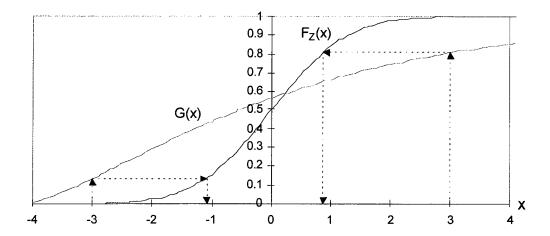


Figure 4. This plot demonstrates the normalizing transformation. F_zX is the standard normal cumulative distribution function, (Cdf), and G(X) is the Cdf of a general distribution. The normalizing function in this case is the one that transforms, for example, -3 to 1.1 and 3 to 0.9.

To see this, suppose f is known, let X be a random variable from a continuous distribution, G, so that f(X) = Z is a standard normal random variable. Let $\{x_1, x_2, ..., x_n\}$ be a sample from G, and let T_Z be the p/C tolerance bound based on $\{f(x_1), f(x_2), ..., f(x_n)\}$, using normal tolerance bound theory. Define Z_p to be the p-quantile of a standard normal distribution. Then by the definition of a p/C tolerance bound, we have

$$C = P(T_Z > q_p)$$

= $P(f^{-1}(T_Z) > f^{-1}(z_p))$

Let $T_X = f^{-1}(T_Z)$ and $q_p^X = f^{-1}(z_p)$, and we have

$$C = P(T_X > q_p^X).$$

If we can show that q_p^X is the p-quantile of the distribution of X; i.e.,

$$\int_{-\infty}^{q_p^X} g(x)dx = p$$

then we have demonstrated that $T_X = f^{-1}(T_Z)$ is a p/C tolerance interval for X. This is easily shown as follows:

$$p = P(Z \le z_p)$$

$$= P(f^{-1}(Z) \le f^{-1}(z_p))$$

$$= P(X \le q_p^X),$$

i.e., q_p^X is the p-quantile of X.

If there is sufficient data to estimate the normalizing function, f, or its inverse, f^{-l} , tolerance bounds for non-normal data can be estimated by $f^{-1}(T_Z)$, where T_Z is the tolerance bound assuming a normal distribution (see Fig. 5). The K-value for a 0.997/0.90 normal tolerance interval is 2.83 for a sample of 1131. This corresponds to 1.05e+07 for a 0.997/0.90 tolerance bound for the gust loads data shown in Fig. 5. This can be converted to k-value by using the sample mean and standard deviation of the gust loads data. For a 0.997/0.90 tolerance bound, estimation of the function f^{-l} in the vicinity of T_Z requires a sample of 1000 or more. For the gust loads, shown in Fig. 3, a fifth-order polynomial approximates f^{-l} well on a normal probability plot in the vicinity of T_Z .

Normal Probability Plot for 1131 Gusts

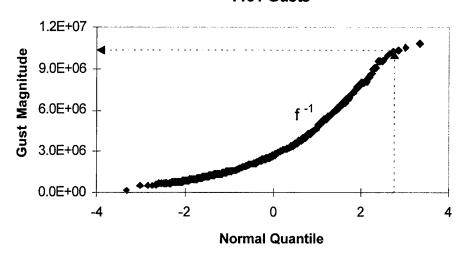


Figure 5. Nonparametric procedure for tolerance bounds. K-value for 0.997/0.90 normal tolerance interval is 2.83 for sample of 1131. From the figure, this corresponds to 1.05e+07 for a 0.997/0.90 tolerance bound for the gust loads data shown. This can be converted to k-value by using the sample mean and standard deviation of the gust.

Applying this method to the data shown in Fig. 1, we obtain a 0.997/0.90 tolerance bound of 11,040,000 inch-pounds.

7. Conclusions

We have presented two methods of finding tolerance bounds for gust loads analysis data. One assumes the gusts follow a gamma distribution, and is useful for smaller sample sizes. The other does not require this assumption, but requires considerably more data. For the data shown in Fig. 1, both methods result in similar tolerance bounds, approximately 11,000,000 inch-pounds for each. The statistical analysis approach presented herein was used to derive the 99.7/90 gust-induced loads in Ref. 3.

References

- Anonymous, "Wind Loads During Ascent," NASA Space Vehicle Design Criteria (Structures), NASA SP-8035, June 1970.
- 2. Kabe, A. M., "Design and Verification of Launch and Space Vehicle Structures," AIAA Paper 98-1718, April 1998.
- 3. Kim, M. C., Kabe, A. M., and Lee, S. S., "Atmospheric Flight Gust Loads Analysis," AIAA Paper 99-1252, April 1999.
- 4. Hahn, G. J., and Meeker, W. Q., Statistical Intervals, A Guide for Practitioners, 1st edition, New York: Wiley, 1991.
- 5. Odeh, R.E., and Owen, D. B., Tables for Normal Tolerance Limits, Sampling Plans, and Screening, 1st edition, New York, Marcel Dekker, Inc., 1980.
- 6. Bain, L. J., and Weeks, D. L., "Tolerance Limits for the Generalized Gamma Distribution," *Journal of the American Statistical Association*, Vol. 60, No. 312, 1965, pp. 1142-1152.
- 7. Spiekerman, C. E., Sako, B. H., and Kabe, A. M., "Identifying Slowly-Varying and Turbulent Wind Features for Day-of-Launch Flight Loads Analyses," AIAA Paper 99-1250, April 1999.
- 8. Smith, O. E., and Adelfang, S. I., "Gust Model Based on the Bivariate Gamma Probability Distribution," *Journal of Spacecraft and Rockets*, Vol. 18, No. 6, 1981, pp. 545-549.
- 9. Spiekermann, C. E., and Kabe, A. M., "Statistical Combination of Launch Vehicle Gust Buffet Atmospheric Flight Loads," AIAA Paper 98-2010, April 1998.
- 10. Efron, B., and Tibshirani, R. J., An Introduction to the Bootstrap, 1st edition, New York, Chapman & Hall, 1993.
- 11. Efron, B., "Better Bootstrap Confidence Intervals," Journal of the American Statistical Association, Vol. 82, No. 397, 1987, pp. 171-185.
- 12. Casella, G., and Berger, R. L., Statistical Inference, 1st edition, Pacific Grove, CA, Wadsworth, 1990, p. 52.



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